

BRAVoS: ROTOR HOVER TEST RIG

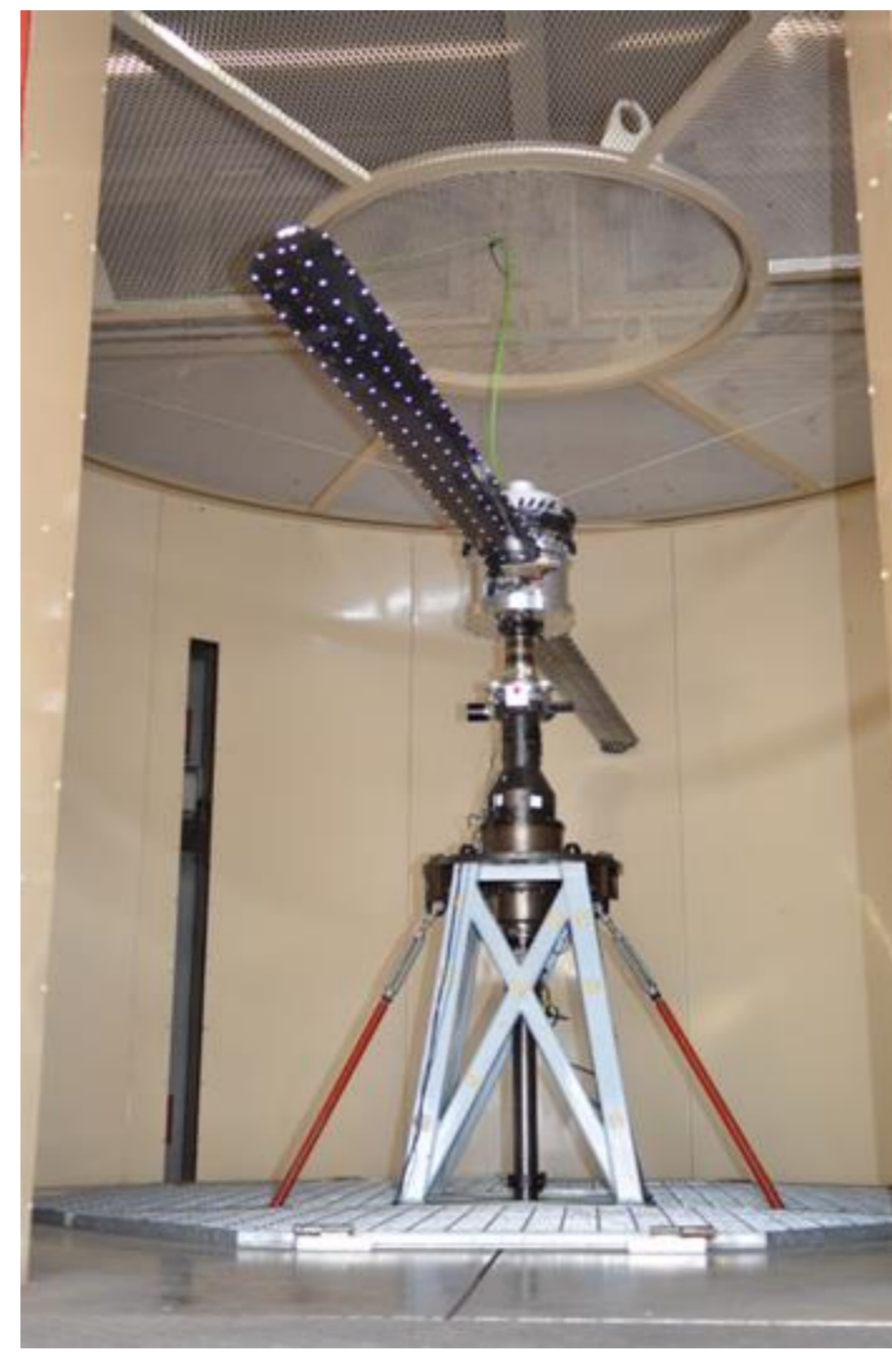
A rotating rig fitted for hover flight with clean internal flow

Major fields of applications: helicopters, wind turbines, propellers, rotor drones

Scope of research: development and validation of experimental measurement methods (rotating blade deflections, ...), characterization of rotating instrumentation under centrifugal loads, aeroelastic stability of hover helicopter rotors, preparation of wind tunnel forward flight rotor tests.

MAIN FEATURES

- Rotor diameter: 4m max
- Rotor speed: 3000 RPM max
- Driving motor Power: 37 KW
- Cyclic pitch management system
 - Akin to real helicopter system
 - Hydraulic powered
 - Remote controlled
- Overall frame configurable for rotor/frame couplings:
 - Rigid locked frame
 - Soft hinged damped frame



Helicopter active two-bladed rotor

MAIN EQUIPMENT

- Camera visualization: rotor tracking, rotating behaviour monitoring
- Telemetry system: 32 channels
- Typical sensors: accelerometers, strain gages, displacement
- Azimuthal position: optical 360 pulses/round

EXPERIMENTAL MEASUREMENT TECHNIQUES

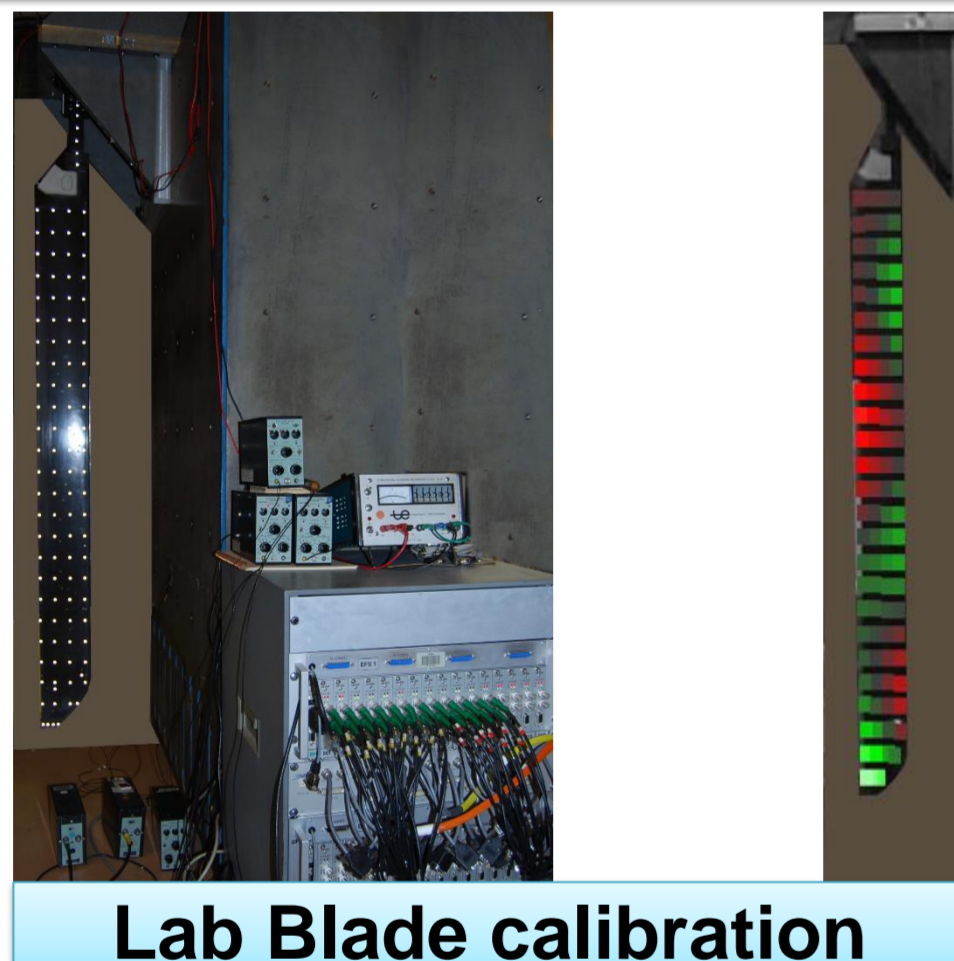
SPA (Strain Pattern Analysis) : from strains to overall deflection displacements

Based on a reduced number of blade embedded strain gauges & a suitable combination of modal shapes

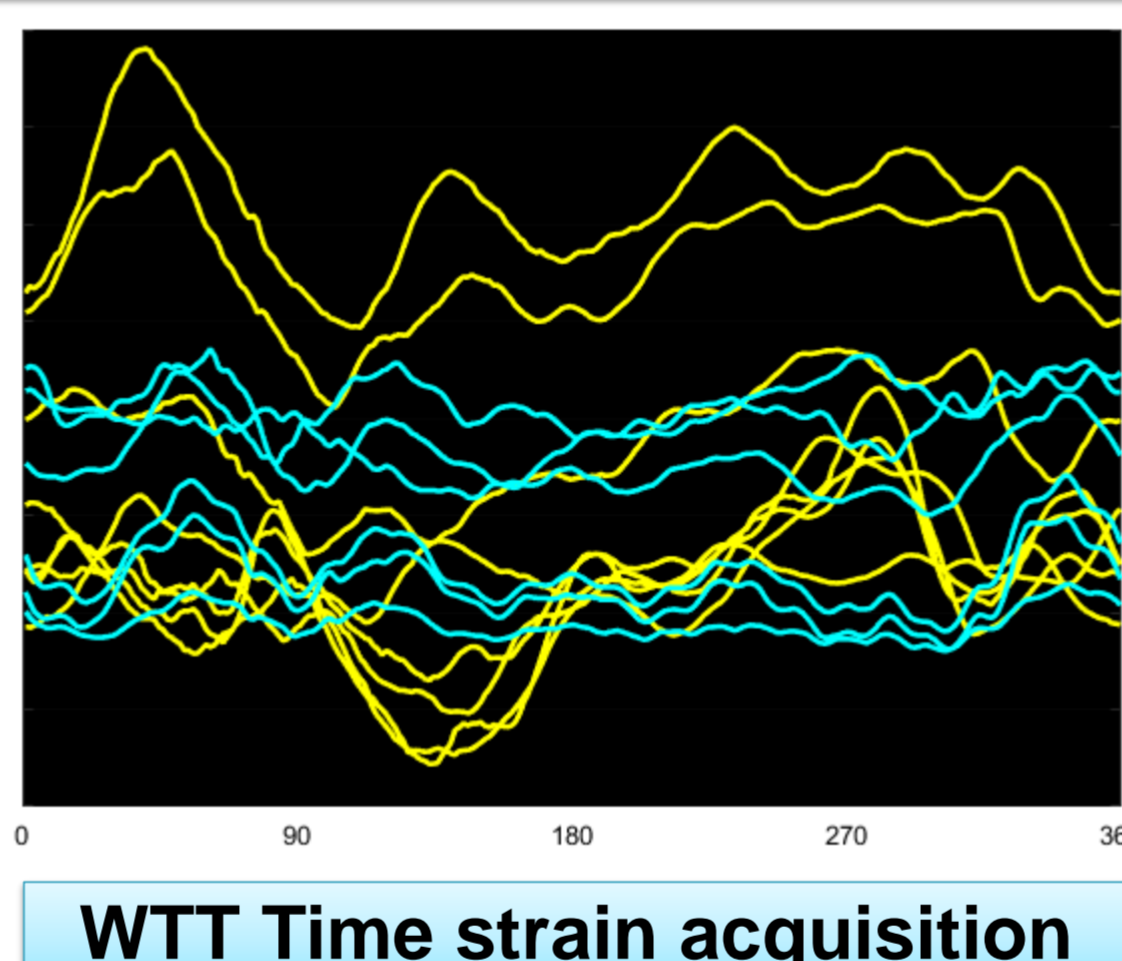
Step 1: Appropriate blade calibration (modal shapes & strains)

Step 2: Setting Transfer matrix T

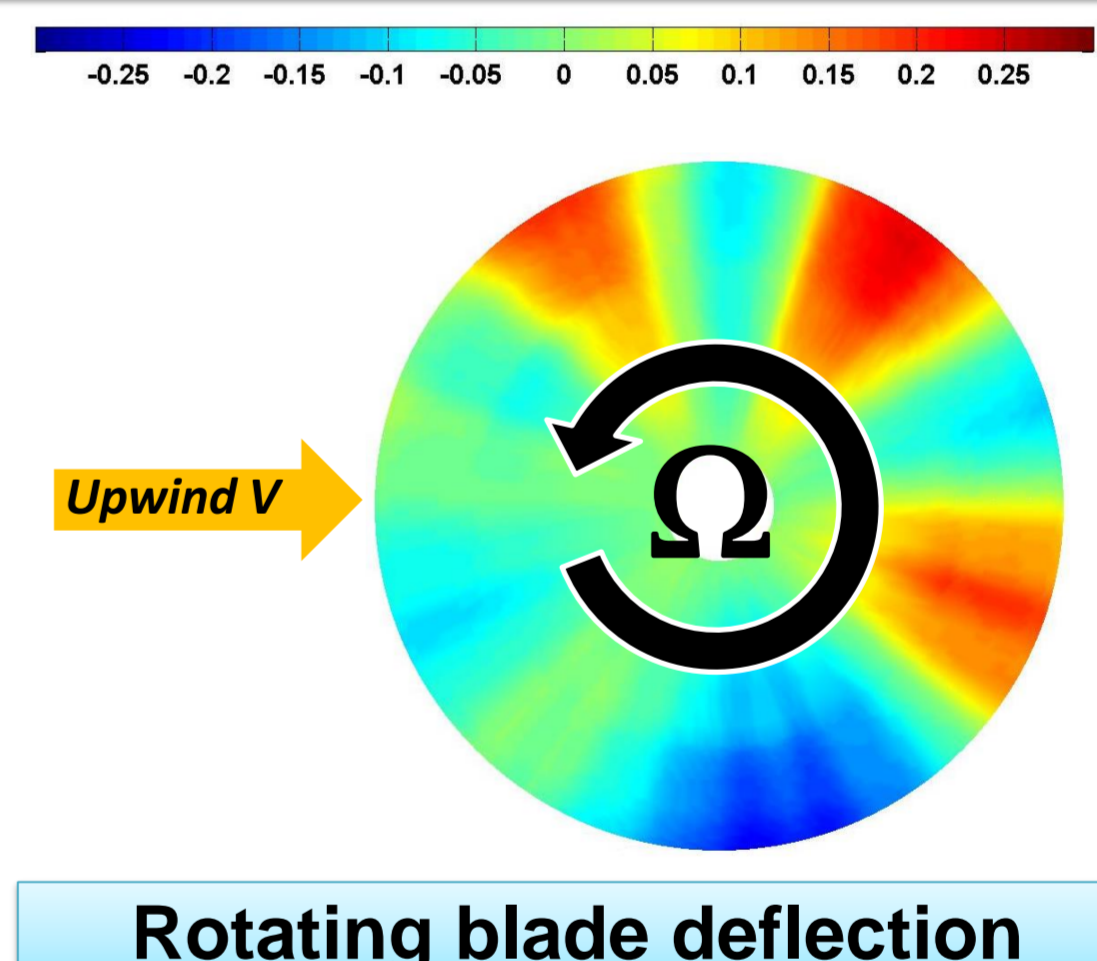
Step 3: Inferring rotating blade deflections Φ from measured rotating strain pattern under load ϵ : $\Phi = T * \epsilon$



Lab Blade calibration



WTT Time strain acquisition



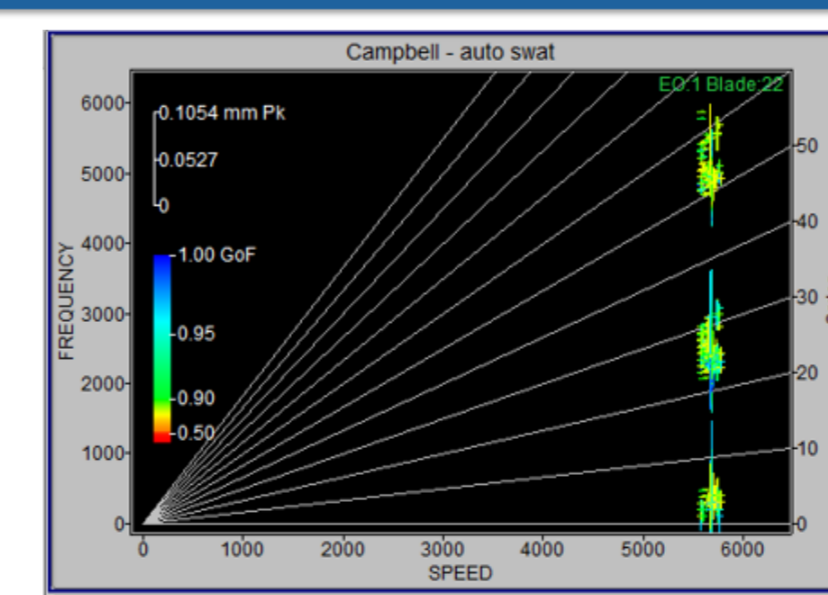
Rotating blade deflection

Tip-timing : from blades passage measurement to blades vibratory characteristics

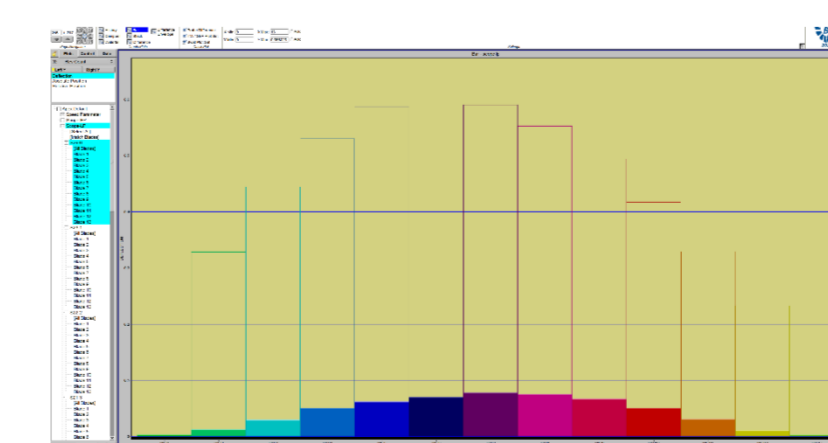
Based on a set of fixed circumferential probes and a high sampling frequency acquisition system

Probes: 18 optical miniaturized laser probes

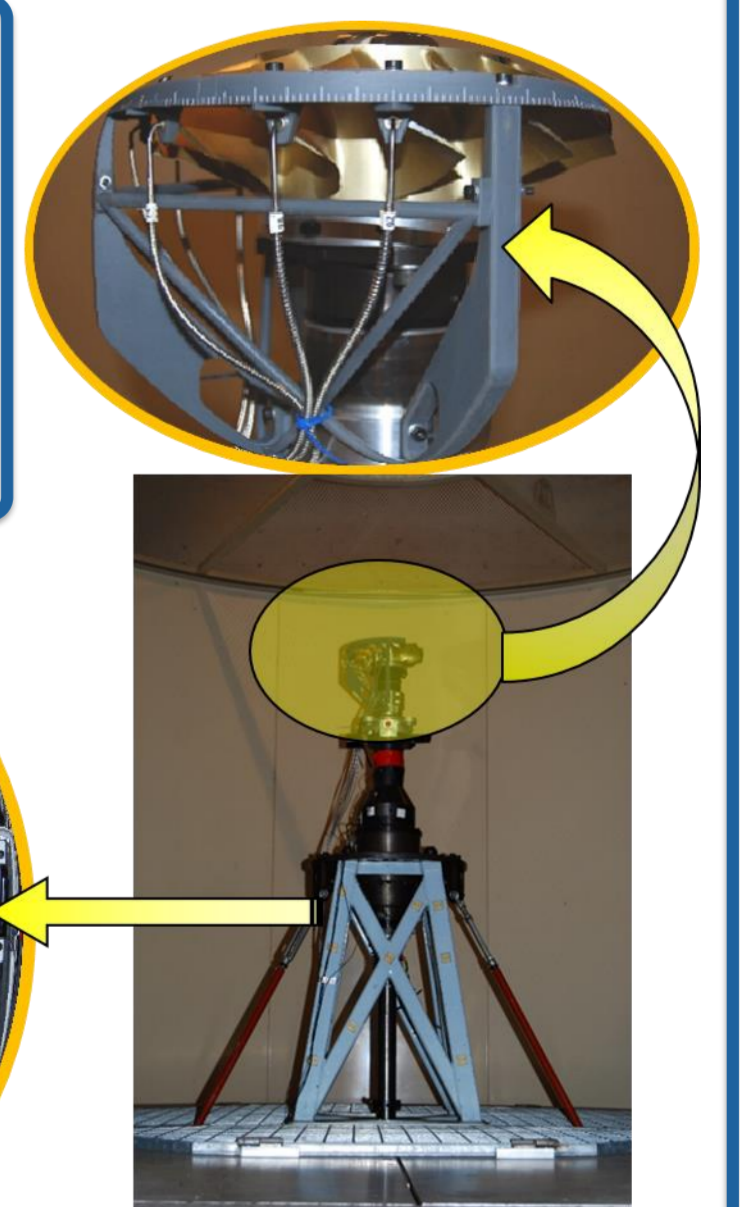
Tip timing acquisition frequency: 100 MHz



Deflection on Campbell



Blades vibratory deflections



Tip-timing system @ BRAVoS

CONTACT

<https://www.onera.fr/en/daaa/contact>