

# RESULTS FROM SIMULATING AND MODELING COMPRESSIBLE TURBULENT FLOWS

*T. B. Gatski*

*thomas.gatski@lea.univ-poitiers.fr*

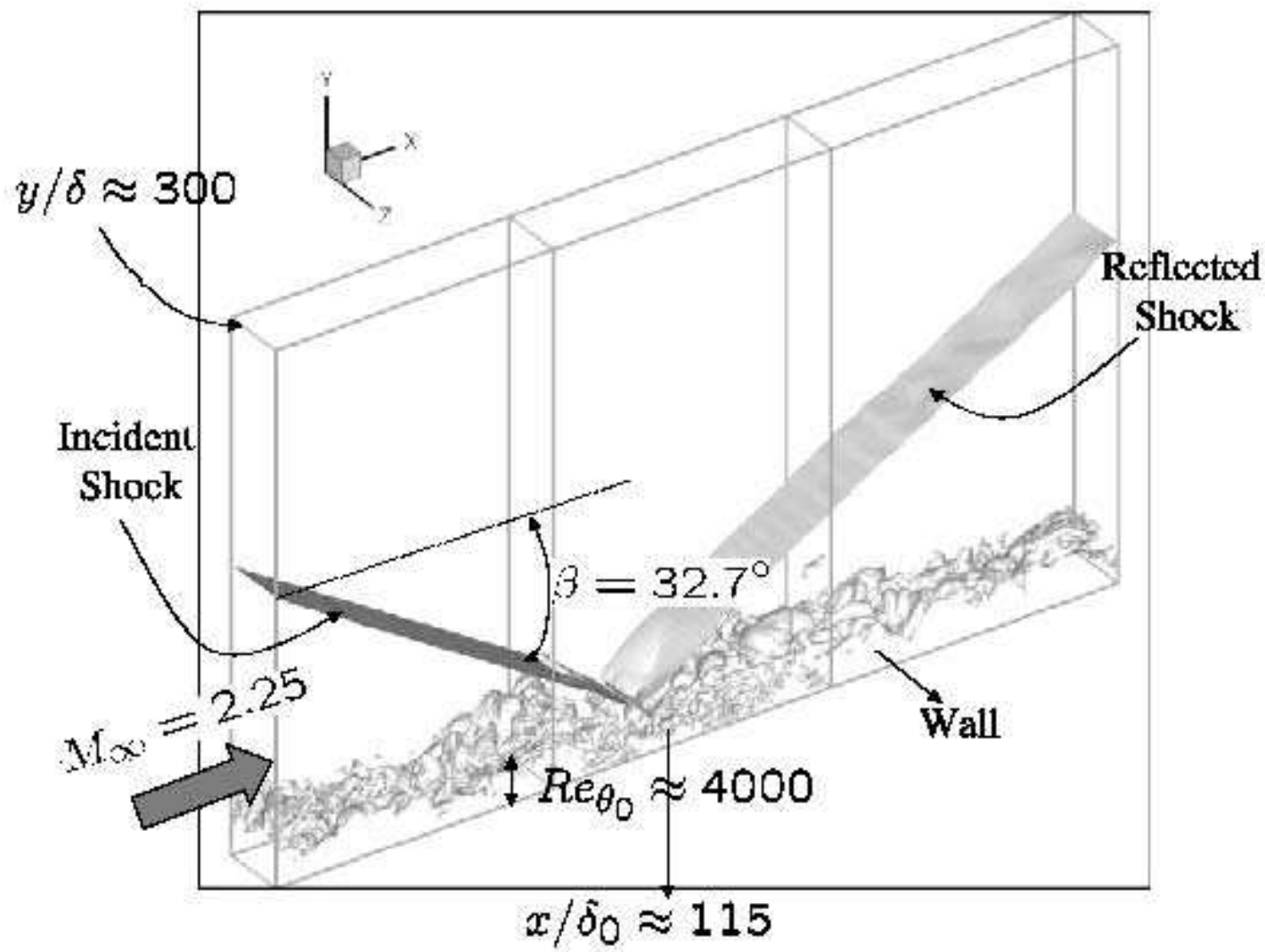
*Laboratoire d'Études Aérodynamiques, UMR CNRS 6609, Université de Poitiers  
BP 30179, 86962 Futuroscope, Chasseneuil Cedex, France*

*Department of Ocean, Earth and Atmospheric Sciences  
Old Dominion University, Norfolk, Virginia 23529 USA*

## DIRECT NUMERICAL SIMULATIONS

- Impinging shock/boundary layer interaction simulation
  - Extends shock-free spatial simulation results
    - *Phys. Fluids, Vol. 16(3), pp. 530-545, 2004*, with S. Pirozzoli and F. Grasso
  - Numerical Algorithm
  - Physical/Computational Domains
  - Mean field simulation results

# DOMAIN PARAMETERS FOR IMPINGING SHOCK CASE



## NUMERICAL ALGORITHM/COMPUTATIONAL DOMAIN

- Full 3D, compressible, unsteady NS equations solved
  - 7th-order WENO discretization of Euler fluxes
  - 4th-order compact discretization of viscous fluxes
  - 4th order RK scheme for time advancement
- Grid resolution:  $2650 \times 111 \times 255$
- Domain size:  $L_x^+ = 100000$ ,  $L_y^+ = 30000$ ,  $L_z^+ = 1.380$
- Grid spacings (interaction zone):  $\Delta_x^+ = 15$ ,  $\Delta_{yw}^+ = 1$ ,  $\Delta_z^+ = 6$

## BOUNDARY CONDITIONS

- Boundary Conditions
  - Inflow: Compressible, Blasius boundary layer solution
  - Outflow:
    - buffer domain used with coarsening in  $x$ -direction
      - first order extrapolation in the supersonic region;
      - uniform pressure in the subsonic sublayer
  - Far-field: nonreflecting boundary conditions

## BOUNDARY CONDITIONS (cont.)

● Wall

$$u = w = 0, \quad v = v_{bs}, \quad T = T_{\infty} \left( 1 + \text{Pr}^{1/3} \frac{\gamma - 1}{2} M_{\infty}^2 \right)$$

$$v_{bs}(x, z, t) = \begin{cases} 0 & x < x_a, \quad x > x_b \\ Au_{\infty} f(x) g(z) h(t) & x_a \leq x \leq x_b \end{cases},$$

$$f(x) = 4 \sin \theta (1 - \cos \theta) / \sqrt{27}, \quad \theta = 2\pi (x - x_a) / (x_b - x_a)$$

$$g(z) = \sum_{l=1}^{l_{max}} Z_l \sin(2\pi l (z/z_{max} + \phi_l)), \quad \sum_{l=1}^{l_{max}} Z_l = 1, \quad Z_l = 1.25 Z_{l+1}$$

$$h(t) = \sum_{m=1}^{m_{max}} V_m \sin(2\pi (\beta t + \phi_m)), \quad \sum_{m=1}^{m_{max}} V_m = 1, \quad V_m = 1.25 V_{m+1}$$

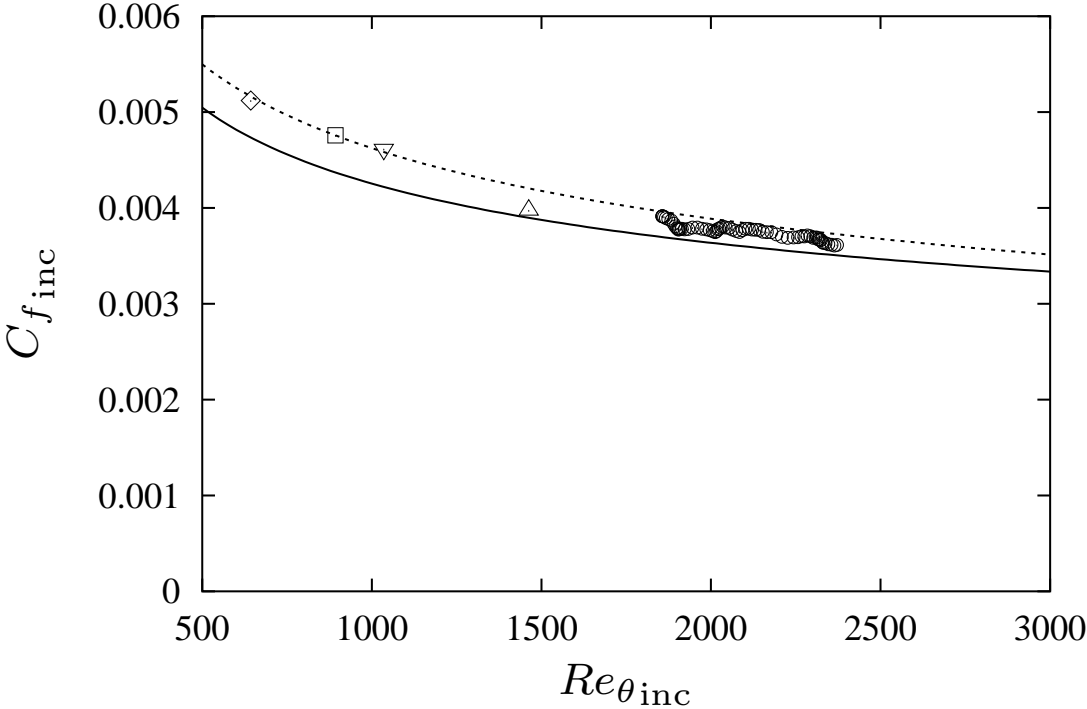
## IMPINGING SHOCK SIMULATION RESULTS

- Streamwise mean pressure variation
- Streamwise skin-friction variation
- Streamwise variation of Van Driest velocity
- Contours in  $x - y$  plane
  - Instantaneous and mean density
  - Instantaneous and mean pressure
  - Instantaneous and mean velocity
  - Turbulent shear stress
  - Turbulent kinetic energy

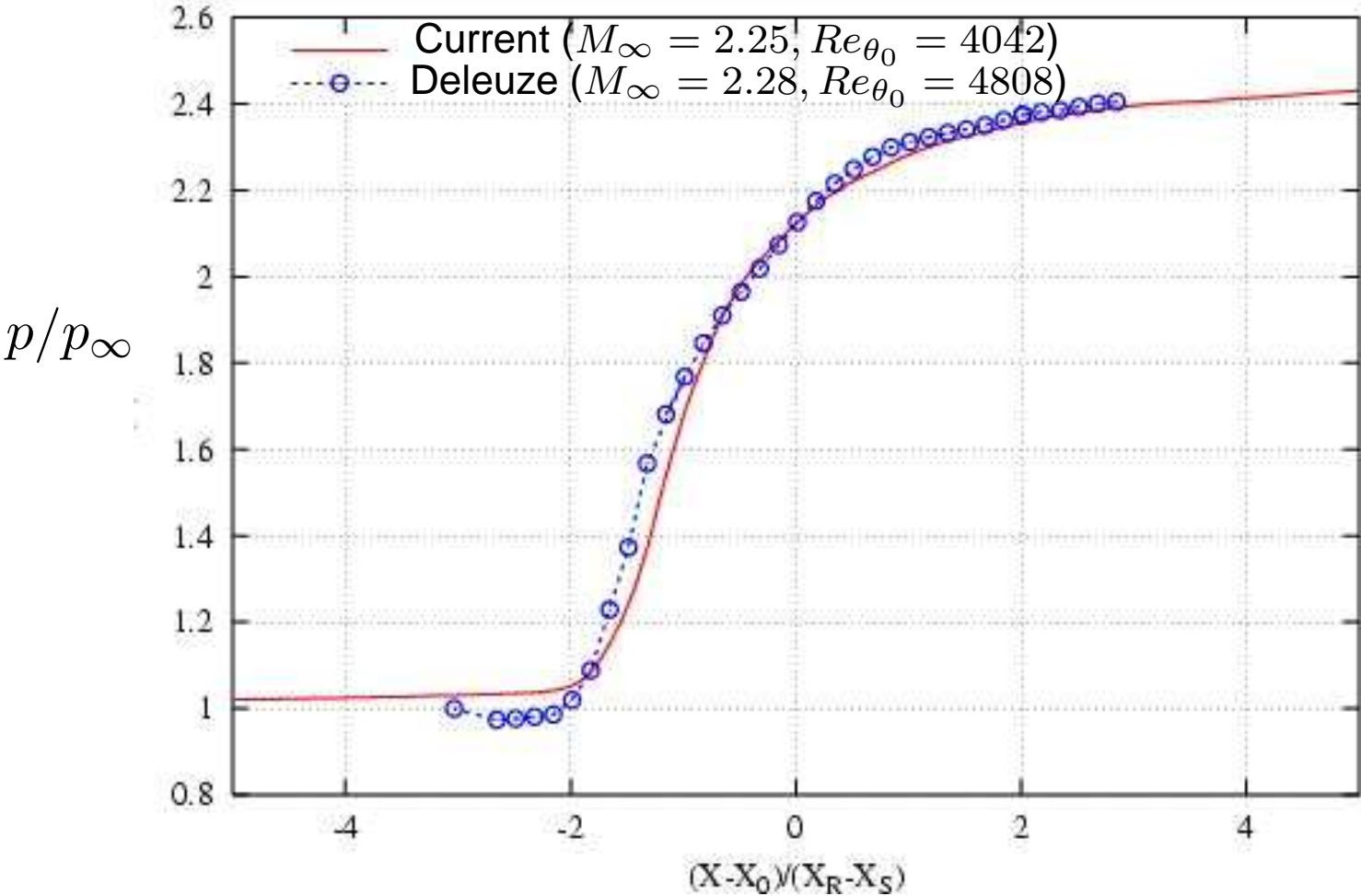
## RELATED DNS STUDIES (WITHOUT SHOCK)

- Guarini, Moser, Shariff and Wray (2000)  
Approximate spatial DNS (under the approximation of slow boundary layer growth)  
Mixed Fourier and B-spline Galerkin method  
 $M_\infty = 2.5, Re_\theta = 1577$     Grid  $256 \times 209 \times 192$
- Maeder, Adams and Kleiser (2001)  
Extended temporal DNS simulation  
Mixed compact-pseudospectral finite-difference method  
 $3 \leq M_\infty \leq 6, Re_\theta \approx 3000$     Grid  $192 \times 144 \times 180$  (at  $M_\infty = 3$ )

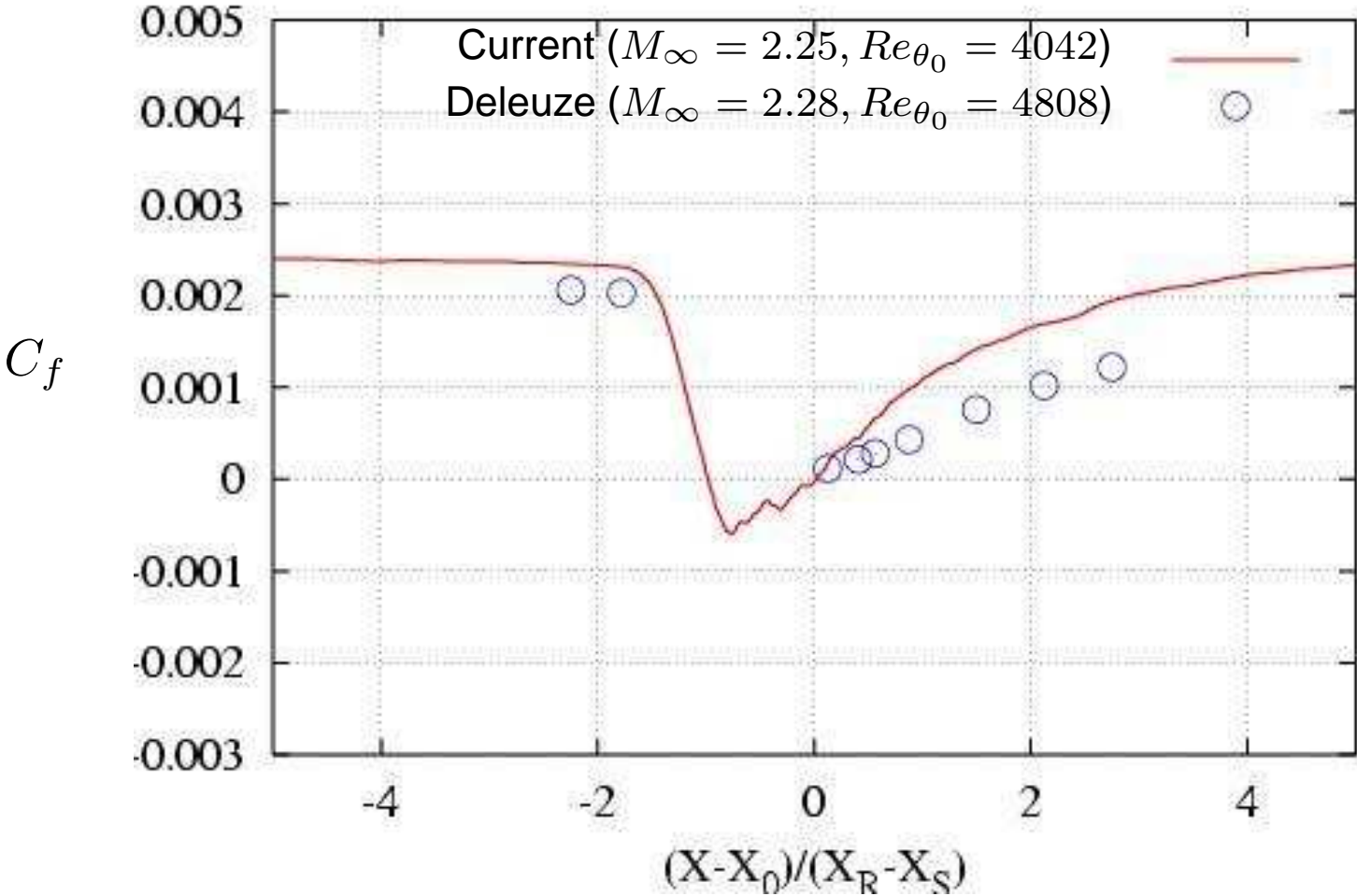
# RELATED DNS STUDIES (WITHOUT SHOCK)



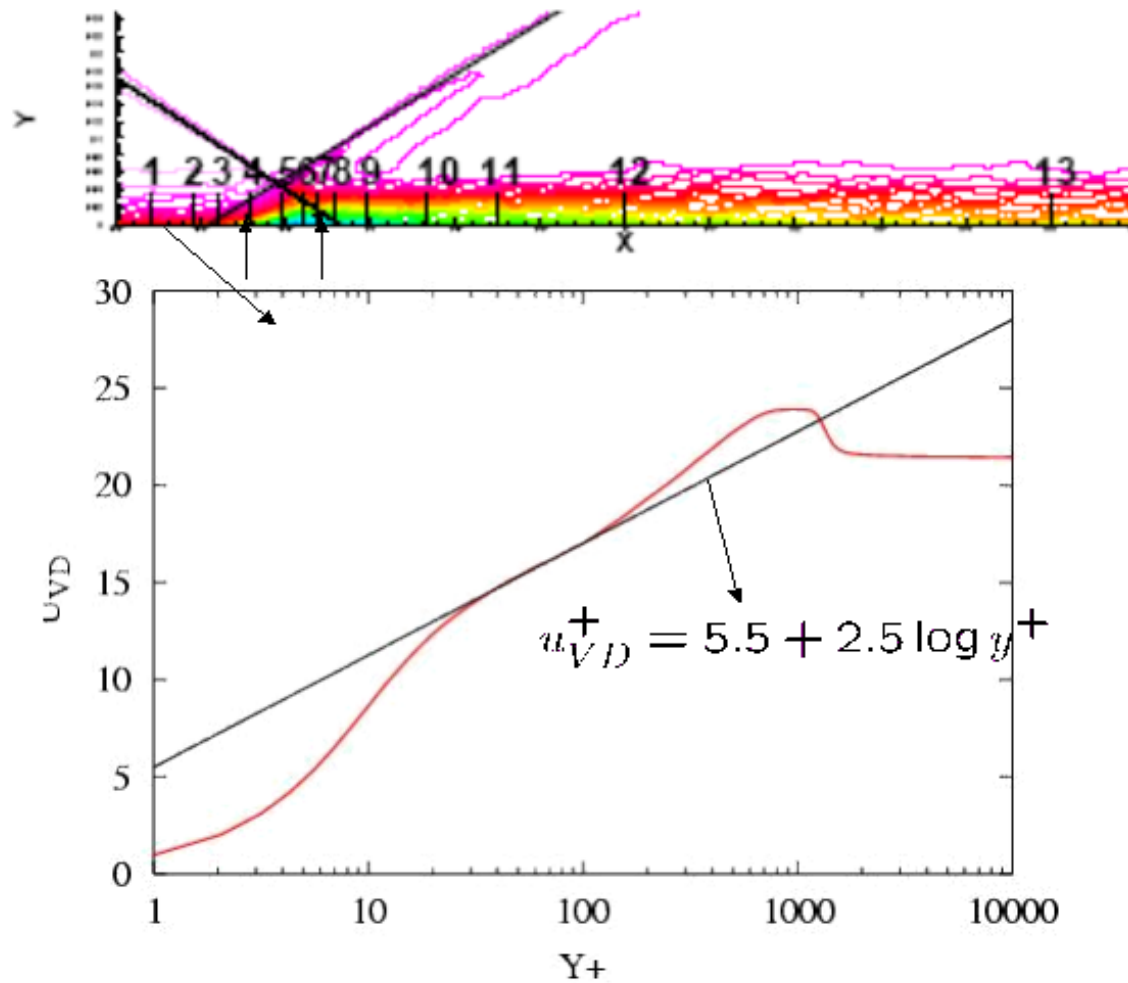
# STREAMWISE MEAN PRESSURE VARIATION



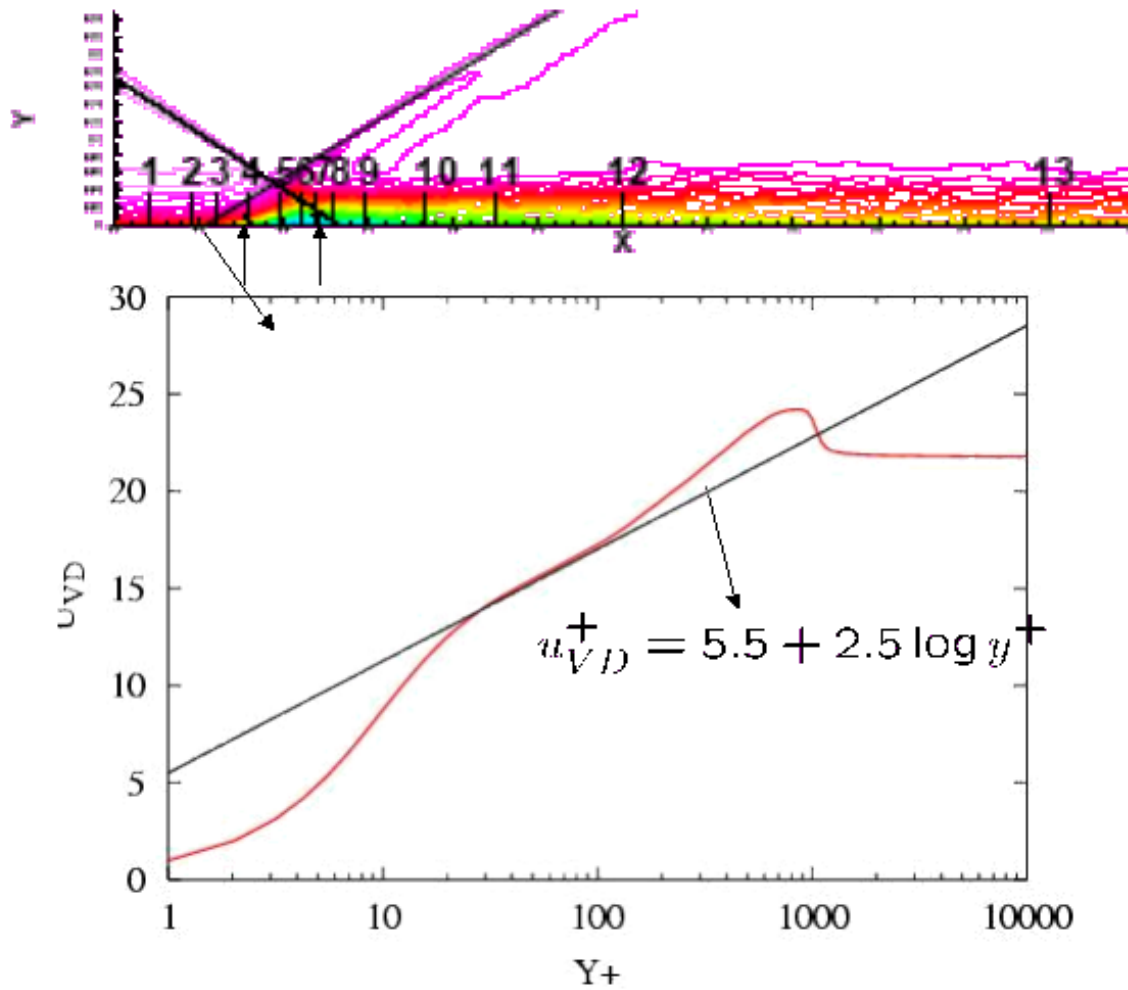
# STREAMWISE SKIN FRICTION VARIATION



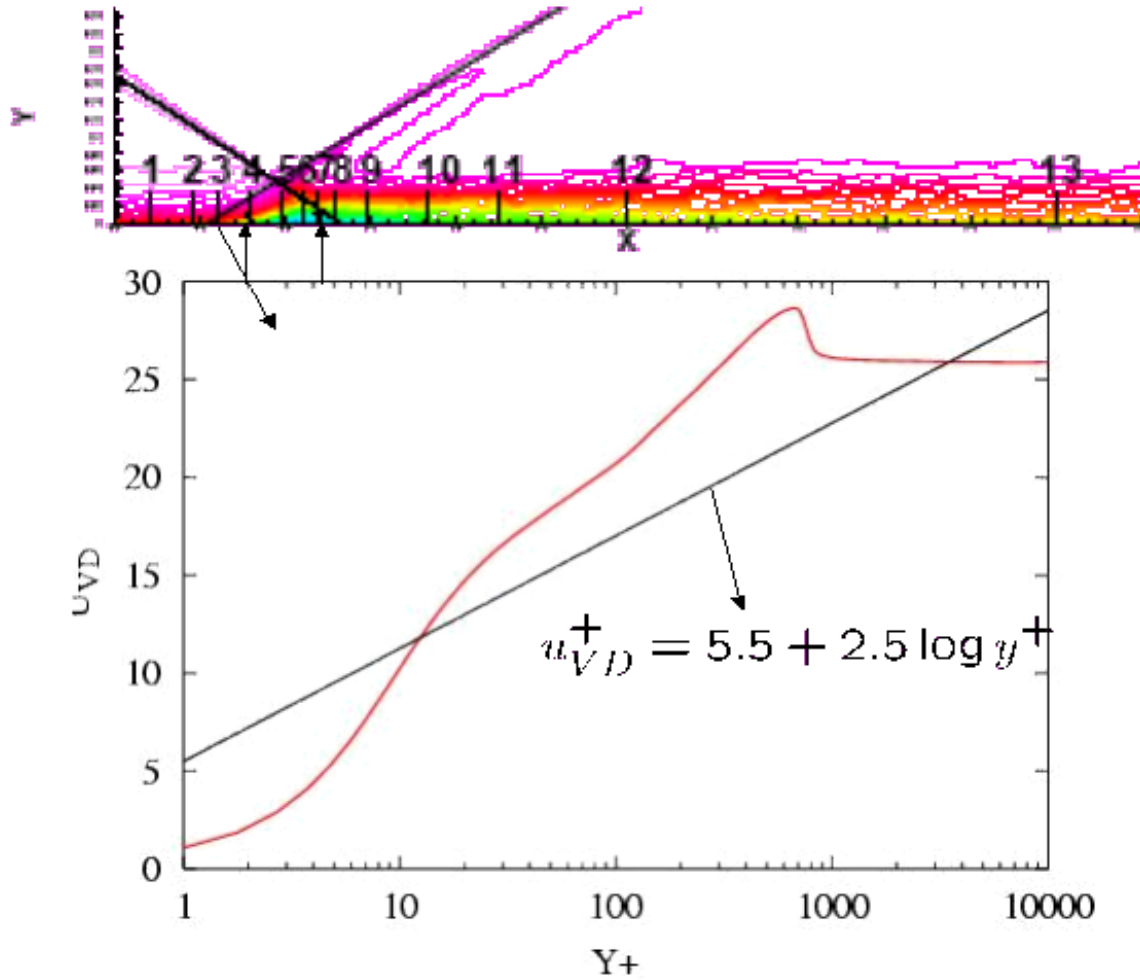
# STREAMWISE VARIATION OF VAN DRIEST VELOCITY



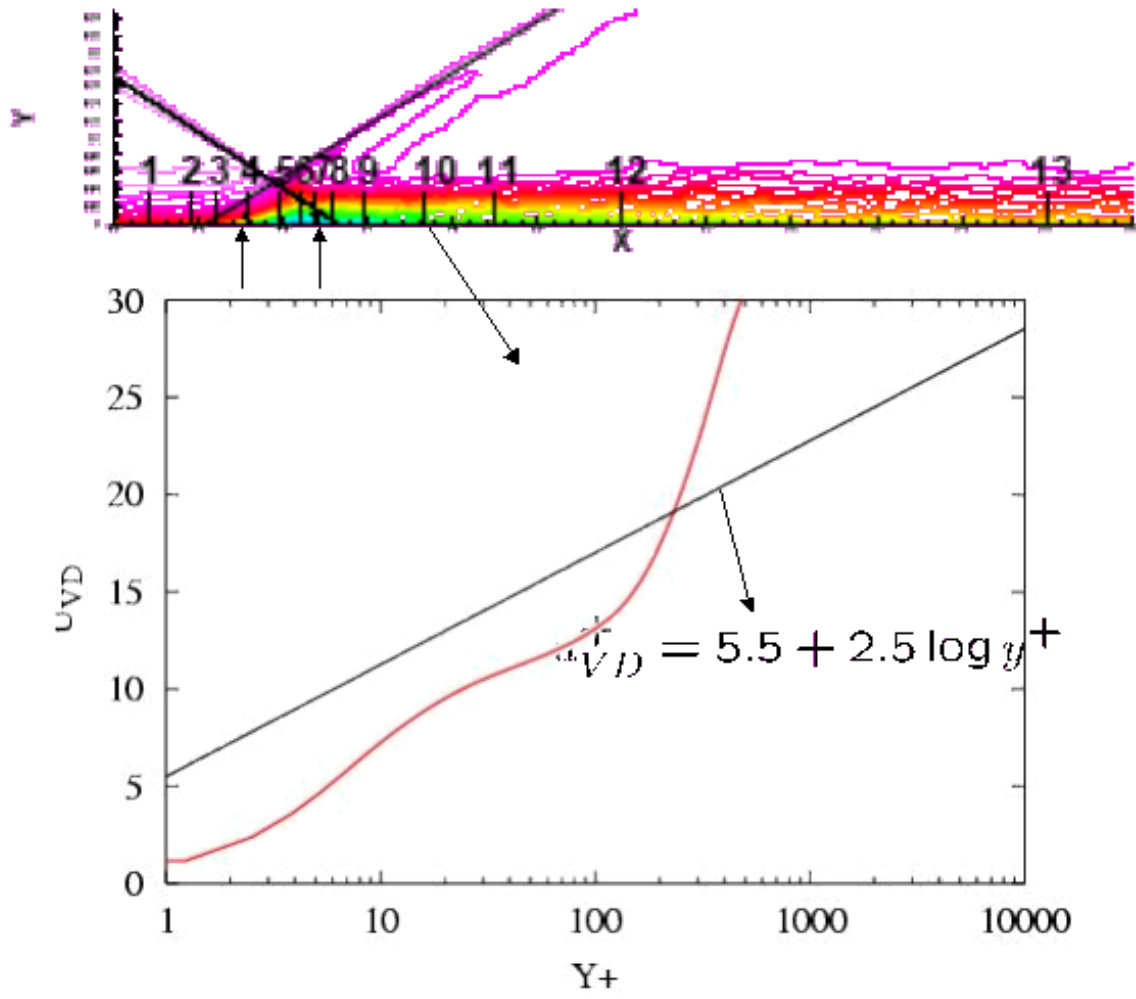
# STREAMWISE VARIATION OF VAN DRIEST VELOCITY



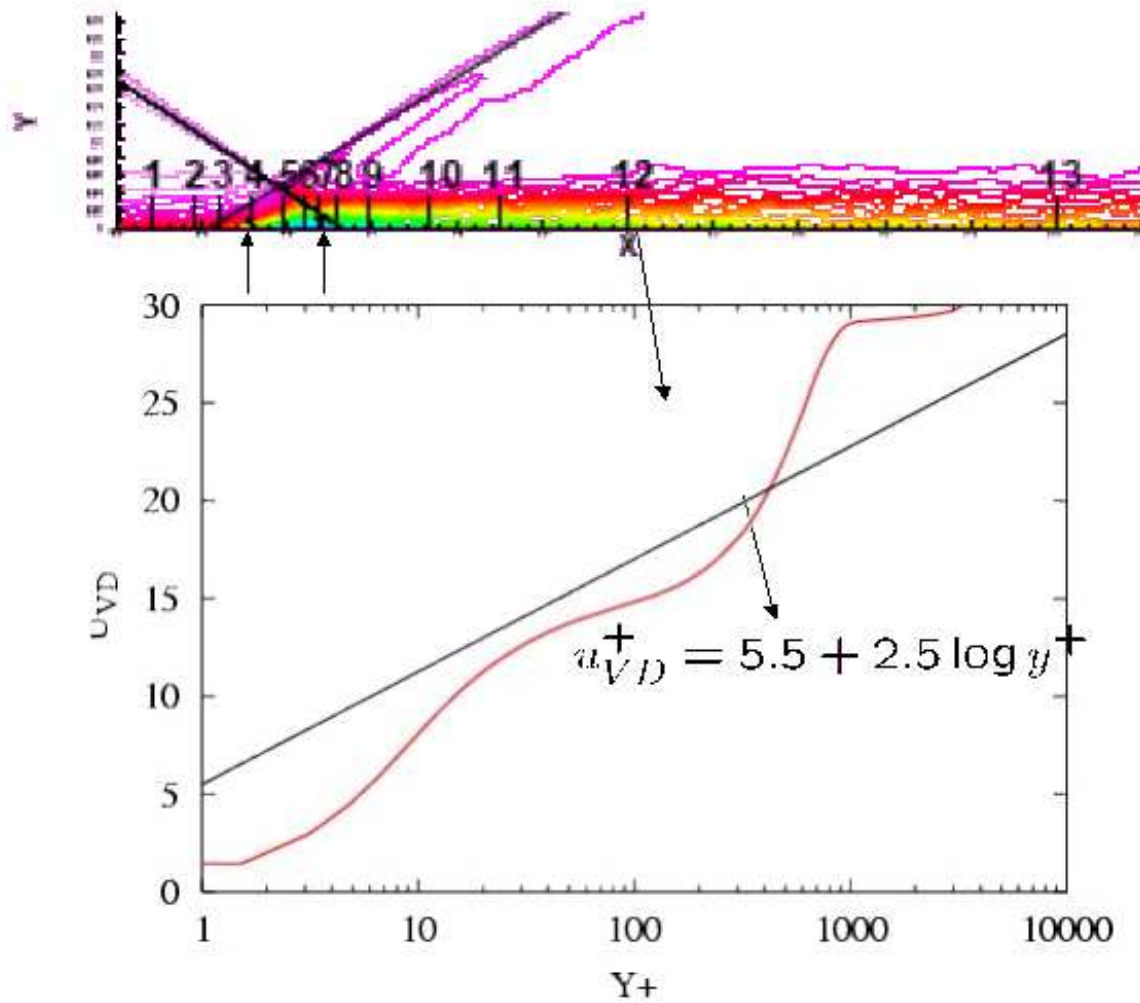
# STREAMWISE VARIATION OF VAN DRIEST VELOCITY



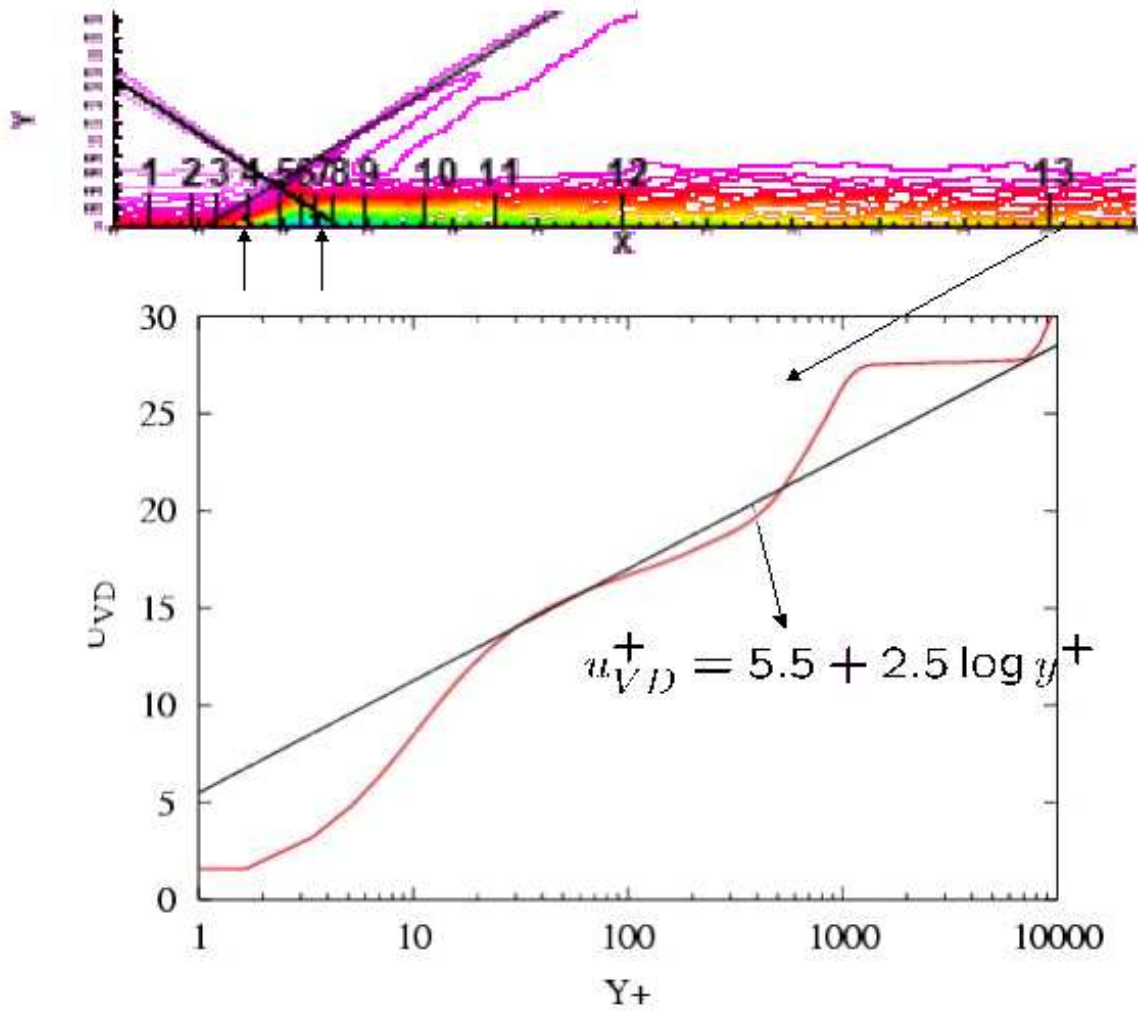
# STREAMWISE VARIATION OF VAN DRIEST VELOCITY



# STREAMWISE VARIATION OF VAN DRIEST VELOCITY



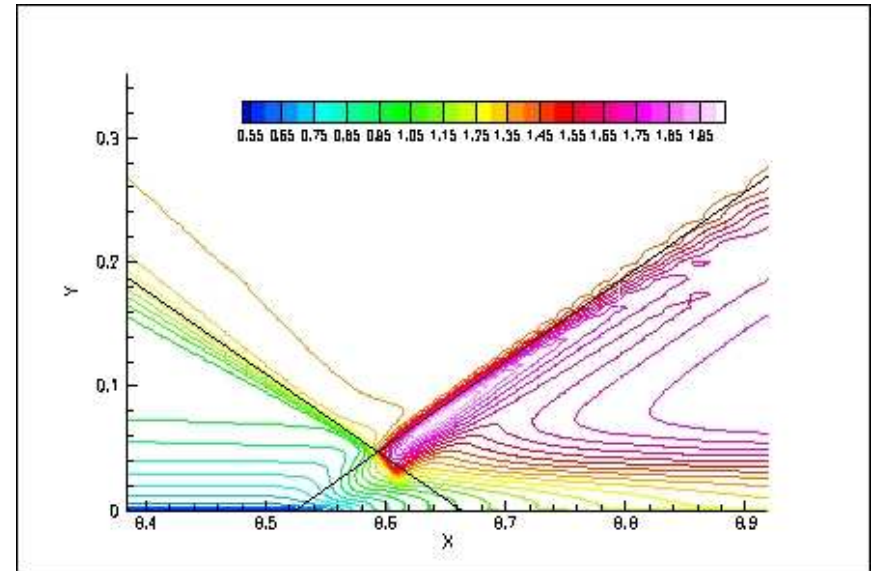
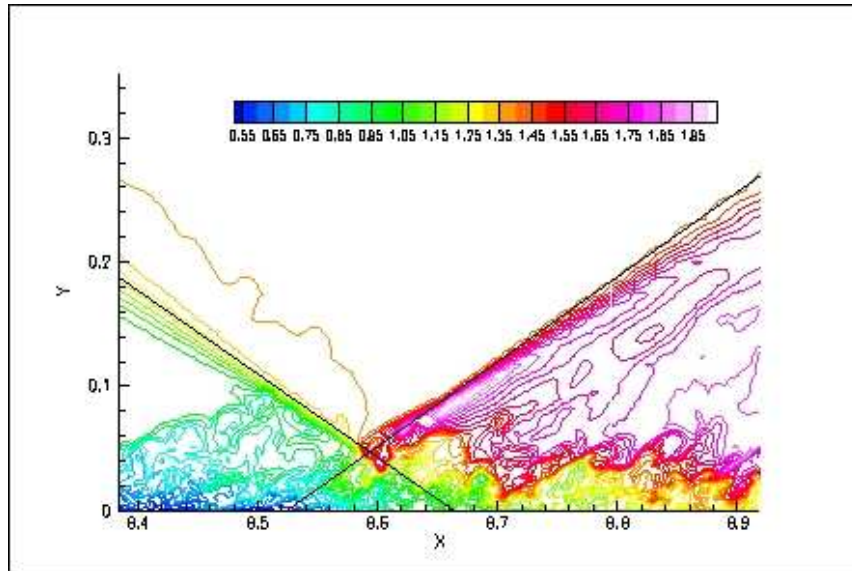
# STREAMWISE VARIATION OF VAN DRIEST VELOCITY



# INSTANTANEOUS AND MEAN DENSITY CONTOURS

$\rho$

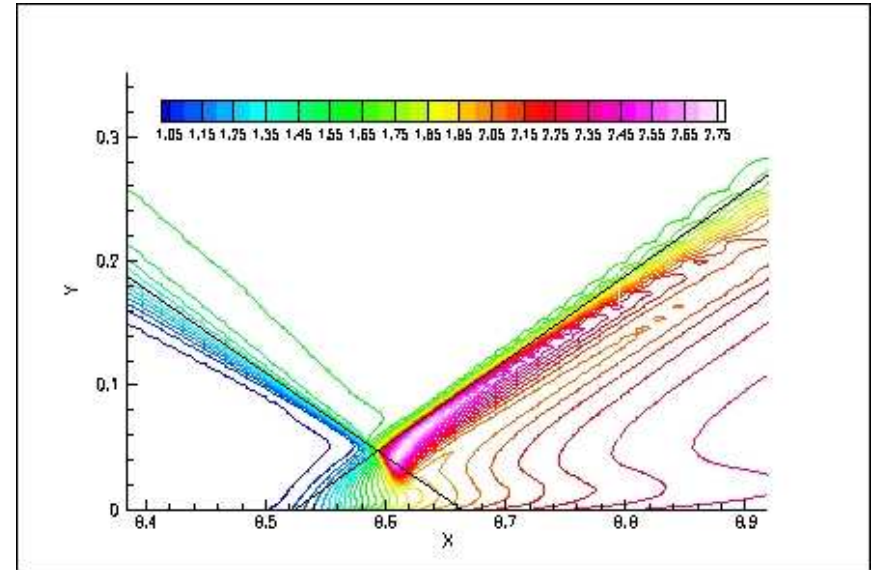
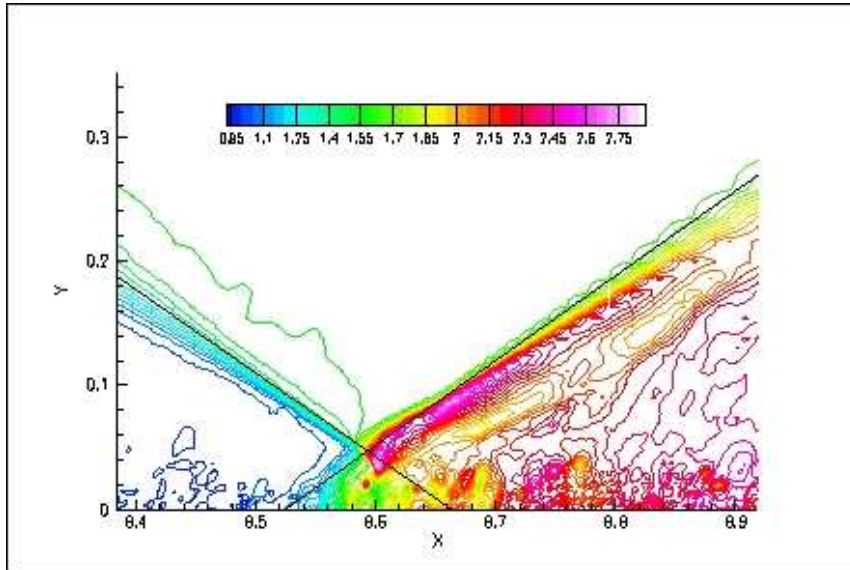
$\bar{\rho}$



# INSTANTANEOUS AND MEAN PRESSURE CONTOURS

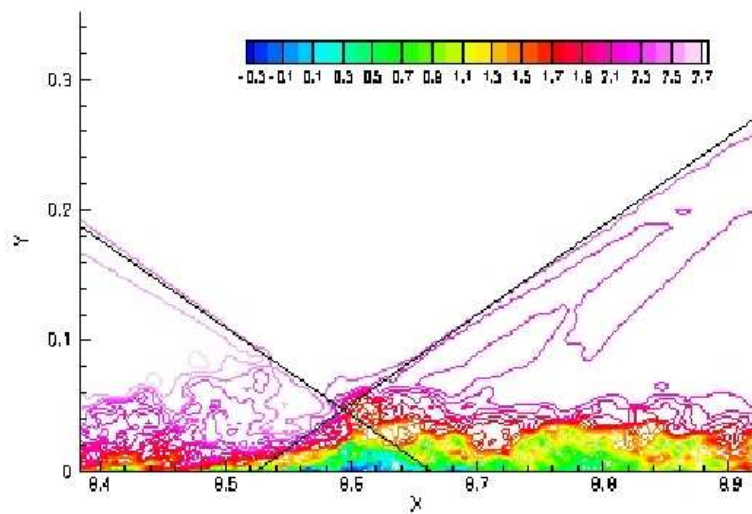
$p$

$\bar{p}$

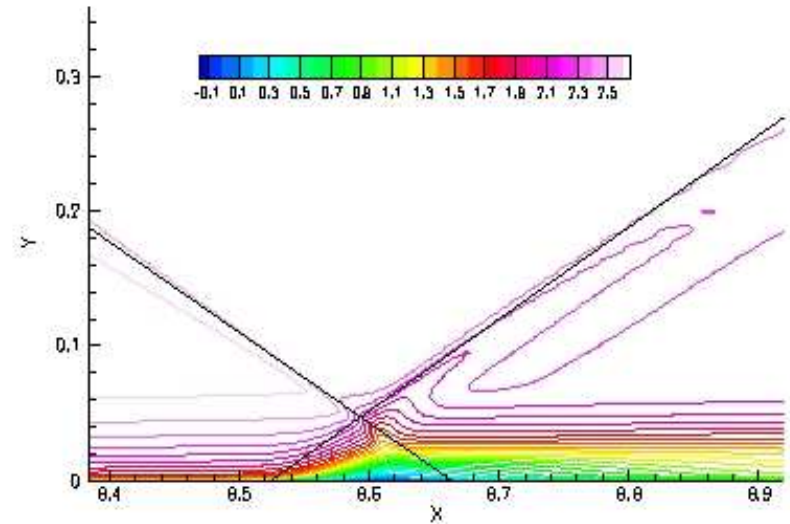


# INSTANTANEOUS AND MEAN VELOCITY CONTOURS

$u$

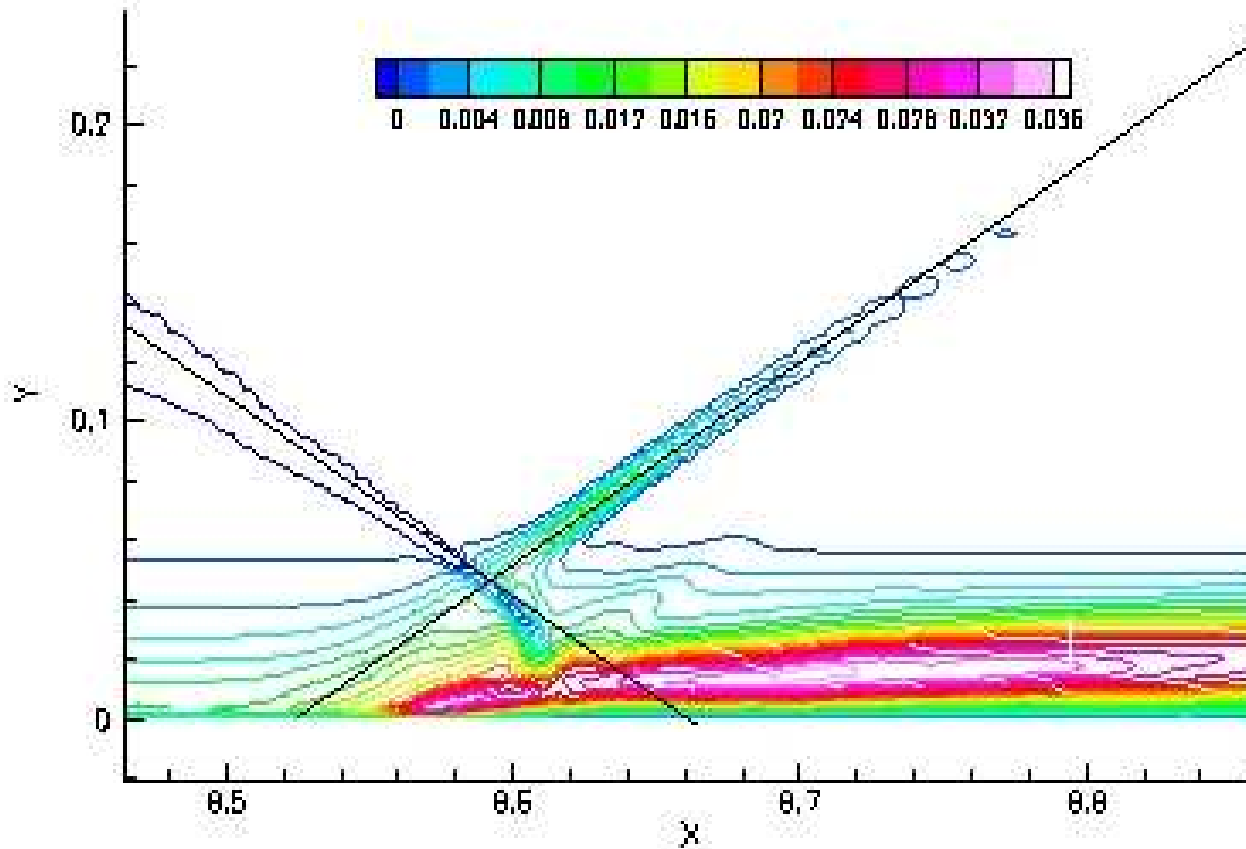


$\tilde{u}$



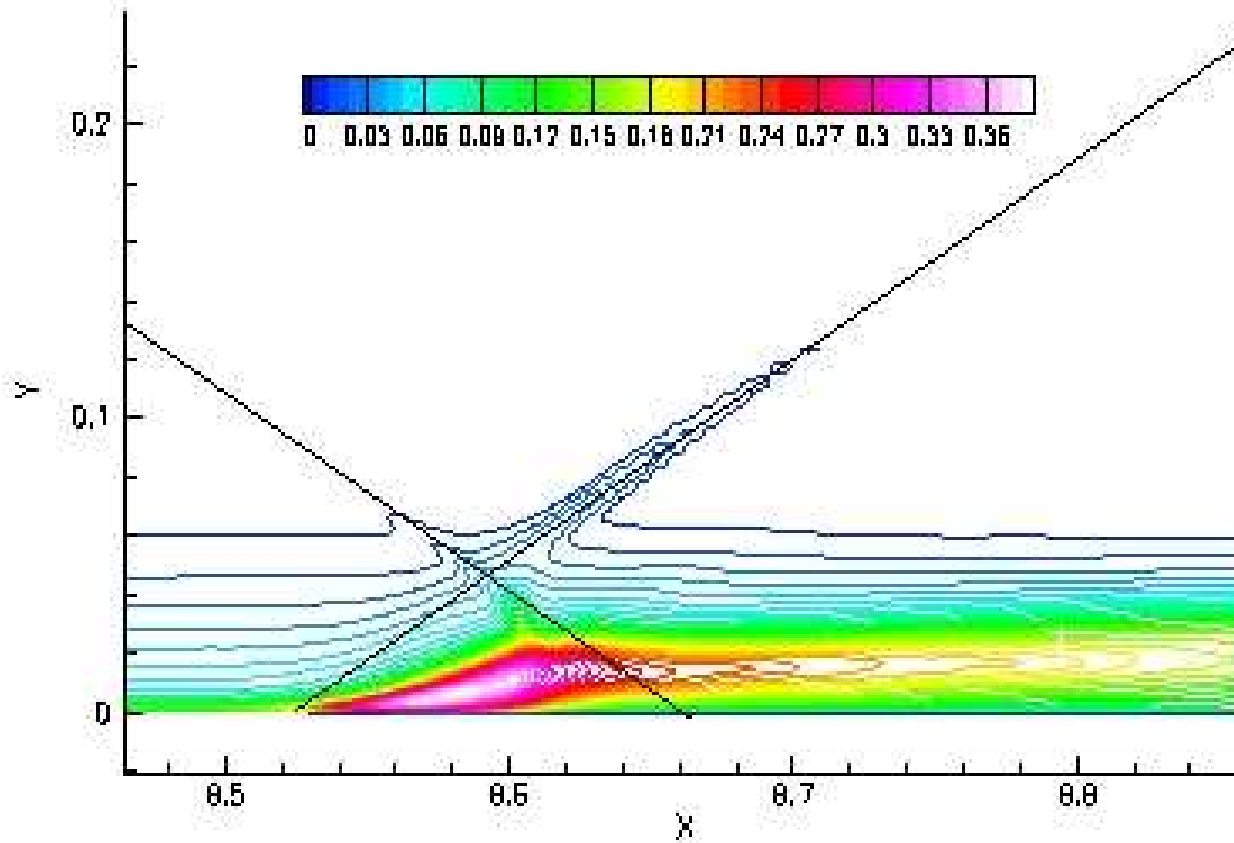
# TURBULENT SHEAR STRESS CONTOURS

$$-\overline{u_1'' u_2''}$$



# TURBULENT KINETIC ENERGY CONTOURS

$$\overline{u_i'' u_i''}$$



## PRESSURE-STRAIN RATE MODELING

- With increasing convective Mach number  $M_c$ , new compressible models should account for
  - Suppression of growth rate of turbulent shear layer
  - Reduction of turbulent kinetic energy of turbulent shear layer
- DNS studies of turbulent kinetic energy budget indicate primary factor is reduction in production term
  - Reduced turbulence production traced to decrease in magnitude of pressure-strain components

## GOVERNING FLUCTUATING EQUATIONS

- Dimensional fluctuating mass and momentum equations

$$\frac{\partial \rho^{*'}}{\partial t^*} + \frac{\partial}{\partial x_j^*} (\rho^{*'} \tilde{u}_j^* + \rho^* u_j^{*''}) = 0$$

$$\begin{aligned} \frac{\partial}{\partial t^*} (\rho^{*'} \tilde{u}_i^* + \rho^* u_i^{*''}) + \frac{\partial}{\partial x_j^*} [\rho^* (\tilde{u}_i^* u_j^{*''} + u_i^{*''} \tilde{u}_j^* + u_i^{*''} u_j^{*''}) \\ - \langle \rho^* \rangle \widetilde{u_i^{*''} u_j^{*''}} + \rho^{*'} \tilde{u}_i^* \tilde{u}_j^*] = - \frac{\partial p^{*'}}{\partial x_i^*} . \end{aligned}$$

- Nondimensionalize with
  - Fluctuating field: length scale  $l_e^*$  and velocity scale  $u_e^*$
  - Mean field: velocity scale  $\mathcal{S}^* l_e^*$
  - Thermodynamic variables: density and pressure scales  $\rho_e^*$  and  $\rho_e^* u_e^{*2}$ , respectively

## WAVE EQUATION FOR FLUCTUATING PRESSURE

$$\left[ \left( M_t \frac{\partial}{\partial t} + M_g \tilde{u}_j \frac{\partial}{\partial x_j} \right)^2 - \frac{\partial^2}{\partial x_j \partial x_j} \right] p' =$$

$$+ 2 \left( \frac{M_g}{M_t} \right) \frac{\partial \tilde{u}_i}{\partial x_j} \frac{\partial (\rho u_j'')}{\partial x_i} + \left( \frac{M_g}{M_t} \right)^2 \frac{\partial \tilde{u}_i}{\partial x_j} \frac{\partial \tilde{u}_j}{\partial x_i} \rho'$$

$$M_t = u_e^* / c^*, \quad M_g = \mathcal{S}^* l_e^* / c^*$$

- Formal solution given in physical and wave-vector space by

$$p'(\mathbf{x}, t) = \int_0^t dt' \int d^3 \mathbf{x}' G(\mathbf{x} - \mathbf{x}', t - t') f(\mathbf{x}', t')$$

$$\hat{p}(\mathbf{k}, t) = \int_0^t dt' \hat{G}(\mathbf{k}, t - t') \hat{f}(\mathbf{k}, t')$$

## GREEN'S FUNCTION FOR HOMOGENEOUS FLOW

- Green's function  $G(\mathbf{x}, t)$  given by

$$G(\mathbf{x}, t) = \frac{1}{(2\pi)^3} \int d^3\mathbf{k} e^{i\mathcal{K}(\mathbf{k}, t) \cdot \mathbf{x}} \tilde{G}(\mathbf{k}, t)$$

where moving wave-vector  $\mathcal{K}$  is given by

$$\mathcal{K}(\mathbf{k}, t) = \mathbf{k} \left[ \mathbf{I} - \left( \frac{M_g}{M_t} t \right) \boldsymbol{\lambda} \right]$$

- Isothermal homogeneous shear flow  $\tilde{u}_i(\mathbf{x}) = \lambda_{ij} x_j$

$$\boldsymbol{\lambda} = [\lambda_{ij}] = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix}$$

## GREEN'S FUNCTION FOR HOMOGENEOUS FLOW

- $\tilde{G}(\mathbf{k}, t)$  satisfies the equation

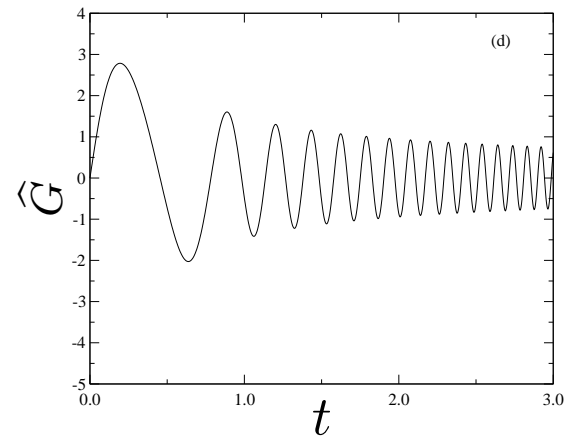
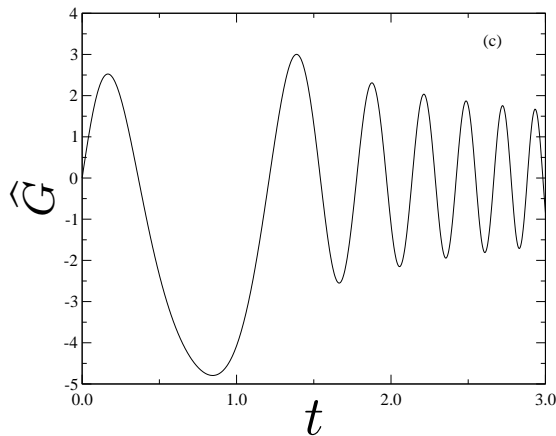
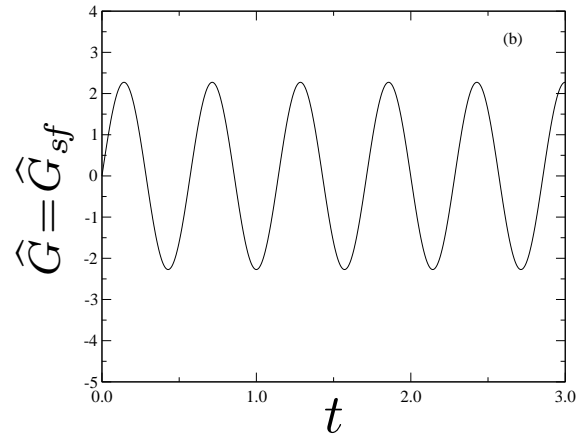
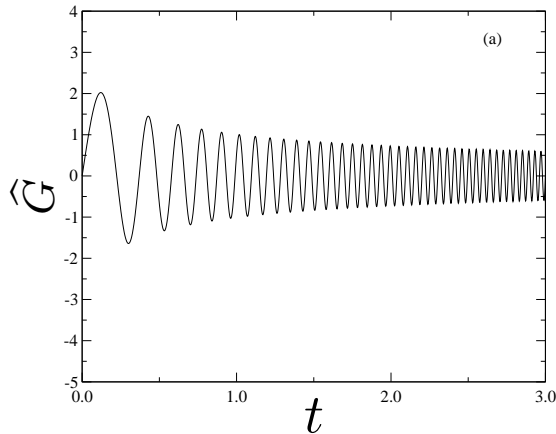
$$\left[ M_t^2 \frac{\partial^2}{\partial t^2} + \left( \frac{M_g k_1}{M_t} \right)^2 \left( t - \frac{k_2 M_t}{k_1 M_g} \right)^2 + (k_1^2 + k_3^2) \right] \tilde{G}(\mathbf{k}, t) = \delta(t)$$

- Limiting case of shear-free flow ( $M_g = 0$ ) studied previously (Pantano and Sarkar, 2002)

$$\tilde{G}_{sf}(k, t) = \frac{\mathcal{H}(t)}{M_t k} \sin(kt/M_t)$$

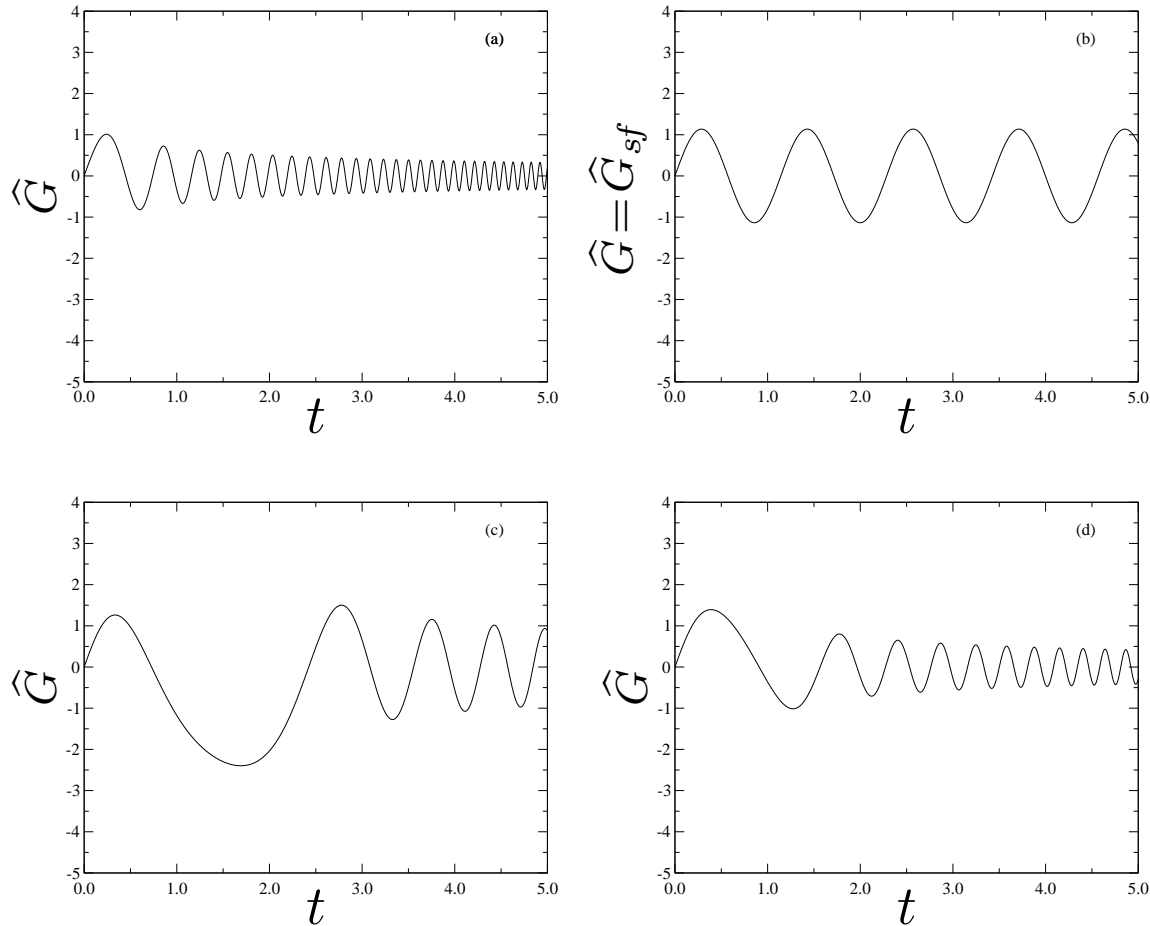
where  $k^2 = k_i k_i$ ,  $\mathcal{H}(t)$  the Heaviside function

## ANISOTROPY OF GREEN'S FUNCTION



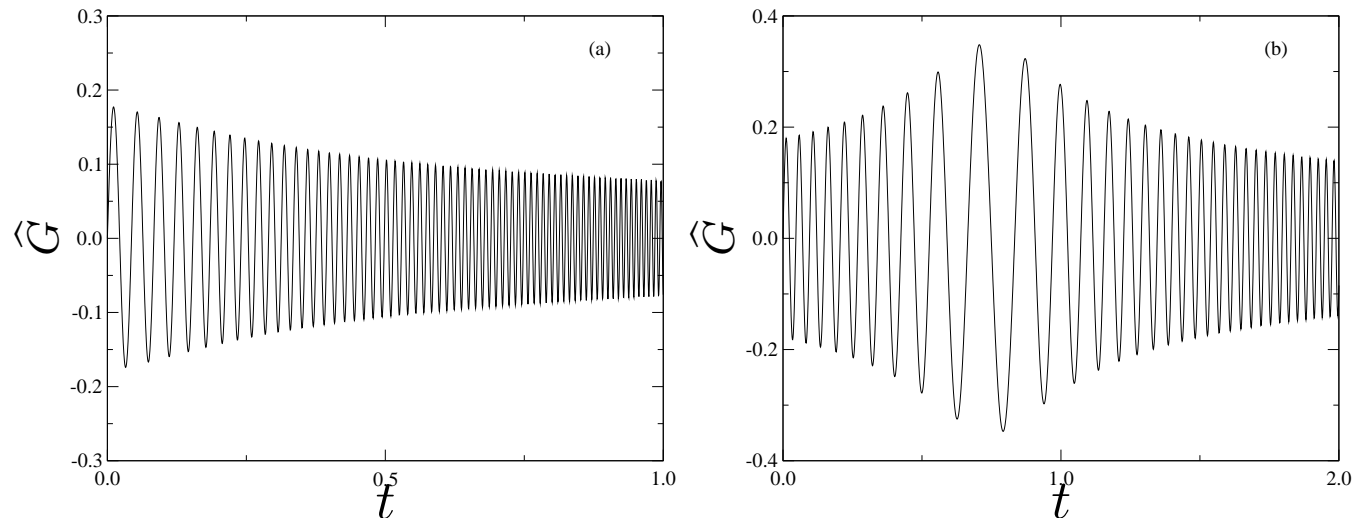
$M_t = 0.2$  and  $M_g = 1.0$ : Green's function at various values of the azimuthal angle  $\phi$  on a circle of radius  $k = 2.2$  in the shear plane  $\theta = \pi/2$  ( $k_3 = 0$ ): (a)  $\phi = \pi/12$ , (b)  $\phi = \pi/2$ , (c)  $\phi = 7\pi/12$ , (d)  $\phi = 2\pi/3$

## ANISOTROPY OF GREEN'S FUNCTION



$M_t = 0.4$  and  $M_g = 1.0$ : Green's function at various values of the azimuthal angle  $\phi$  on a circle of radius  $k = 2.2$  in the shear plane  $\theta = \pi/2$  ( $k_3 = 0$ ): (a)  $\phi = \pi/12$ , (b)  $\phi = \pi/2$ , (c)  $\phi = 7\pi/12$ , (d)  $\phi = 2\pi/3$

## ANISOTROPY OF GREEN'S FUNCTION



$M_t = 0.2$  and  $M_g = 1.0$ : Green's function on circle of radius  $k = 28.0$  in the shear plane  $\theta = \pi/2$  ( $k_3 = 0$ ) with wave-vector azimuthal angle: (a)  $\phi = \pi/12$ , (b)  $\phi = 7\pi/12$

## PRESSURE-STRAIN RATE CORRELATION

- With determination of Green's function, now possible to obtain solution to convective wave equation for (rapid part) of fluctuating pressure
- Expression for two-point pressure-strain rate correlation  $\Pi_{ij}$ ,

$$\Pi_{ij}(\mathbf{r}, t) = \left\langle p'(\mathbf{x} + \mathbf{r}, t) \left( \frac{\partial}{\partial x_j} u_i''(\mathbf{x}, t) + \frac{\partial}{\partial x_i} u_j''(\mathbf{x}, t) \right) \right\rangle$$

$$\begin{aligned} \Pi_{ij}(\mathbf{r}, t) = & \frac{\iota}{\mathcal{V}} \int d^3\mathbf{x} \int d^3\mathbf{k}' d^3\mathbf{k}'' e^{\iota\mathbf{k}' \cdot (\mathbf{x} + \mathbf{r})} e^{\iota\mathbf{k}'' \cdot \mathbf{x}} \\ & \times \langle \hat{p}(\mathbf{k}', t) (k_j'' \hat{u}_i(\mathbf{k}'', t) + k_i'' \hat{u}_j(\mathbf{k}'', t)) \rangle \end{aligned}$$

## OTHER FLUCTUATING PRESSURE CORRELATIONS

- Expression for two-point pressure-velocity correlation,

$$\begin{aligned} \left\langle \overline{p'(\mathbf{x} + \mathbf{r}, t) u_i''(\mathbf{x}, t)} \right\rangle &= \frac{\iota}{\mathcal{V}} \int d^3 \mathbf{x} \int d^3 \mathbf{k}' d^3 \mathbf{k}'' e^{\iota \mathbf{k}' \cdot (\mathbf{x} + \mathbf{r})} e^{\iota \mathbf{k}'' \cdot \mathbf{x}} \\ &\quad \times \left\langle \hat{p}(\mathbf{k}', t) \hat{u}_i(\mathbf{k}'', t) \right\rangle \end{aligned}$$

- Expression for two-point pressure correlation

$$\begin{aligned} \left\langle \overline{p'(\mathbf{x}, t) p'(\mathbf{x} + \mathbf{r}, t)} \right\rangle &= \frac{\iota}{\mathcal{V}} \int d^3 \mathbf{x} \int d^3 \mathbf{k}' d^3 \mathbf{k} e^{\iota \mathbf{k}' \cdot (\mathbf{x} + \mathbf{r})} e^{\iota \mathbf{k} \cdot \mathbf{x}} \\ &\quad \times \left\langle \hat{p}(\mathbf{k}', t) \hat{p}(\mathbf{k}, t) \right\rangle \end{aligned}$$

## PRESSURE-STRAIN RATE CORRELATION

- In terms of the energy spectrum tensor

$$\Pi_{ij} = 2\bar{\rho} \left( \frac{M_g}{M_t} \right) \int_0^\infty dk \int_0^\pi \sin \theta d\theta \int_0^{2\pi} d\phi R(k, \theta, \phi; \tau_I) \\ \times k_l \lambda_{lm} [E_{mi}(k, \theta, \phi) k_j + E_{mj}(k, \theta, \phi) k_i]$$

- $R(k, \theta, \phi; \tau_I)$  dependent on the anisotropic Green's function and temporal decorrelation time  $\tau_I$ 
  - For incompressible flow:  $R(k, \theta, \phi; \tau_I)$  not present
  - For compressible, shear-free flow:

$$R(k, \theta, \phi; \tau_I) = \left( 1 + \frac{M_t^2}{\tau_I^2 k^2} \right)^{-1}$$

- Representation for energy spectrum tensor  $E_{ij}$  needed

## ENERGY SPECTRUM TENSOR

- Linear representation for  $E_{ij}(\mathbf{k}; \mathbf{b})$ 
  - Basis  $\delta_{ij}$ ,  $\hat{k}_i \hat{k}_j$ ,  $b_{ij}$ , and  $(b_{in} \hat{k}_n \hat{k}_j + \hat{k}_i \hat{k}_n b_{nj})$

$$\hat{k}_i = k_i/k$$

$$b_{ij} \equiv \frac{\langle u_i'' u_j'' \rangle}{2K} - \frac{\delta_{ij}}{3}$$

- Constraints: (i) trace of  $E_{ij}$  is proportional to isotropic energy spectral density  $E(k)$  ( $E_{ii} = E(k)/2\pi k^2$ ); (ii) continuity  $k_i E_{ij} = 0$

## ENERGY SPECTRUM TENSOR

- Four-term representation

$$E_{ij}(\mathbf{k}; \mathbf{b}) = \frac{E(k)}{4\pi k^2} \left( \delta_{ij} - \hat{k}_i \hat{k}_j \right) + \frac{E_a(k)}{8\pi k^2} \left( \hat{k}_n b_{nm} \hat{k}_m \right) \left( \delta_{ij} + \hat{k}_i \hat{k}_j \right) + \frac{E_a(k)}{4\pi k^2} \left[ b_{ij} - \left( b_{in} \hat{k}_n \hat{k}_j + \hat{k}_i \hat{k}_n b_{nj} \right) \right]$$

- Expansion coefficient  $E_a(k)$  is anisotropic energy spectral density

## ENERGY SPECTRUM TENSOR

- For incompressible flow ( $R(k, \theta, \phi)$  not present)
  - Shih, Reynolds, Mansour (1990) used a similar expression to obtain a closure model for the LRR pressure-strain rate correlation
  - Exact forms for  $E(k)$  and  $E_a(k)$  not required
  - Obtained from conditions imposed by integral relation between the Reynolds stress energy spectrum and Reynolds stress anisotropy tensor

$$4\pi \int_0^{\infty} k^2 E_{ij} dk = 2K \left( b_{ij} + \frac{\delta_{ij}}{3} \right)$$

## ENERGY SPECTRAL DENSITIES

- For compressible flow ( $R(k, \theta, \phi)$  present)
- Modeled forms for both  $E(k)$  and  $E_a(k)$  required
  - Isotropic energy spectral density

$$E(k) = C\varepsilon^{2/3}k^{-5/3} \left[ \frac{kL}{[(kL)^2 + C_L]^{1/2}} \right]^{17/3}$$

- Anisotropic energy spectral density

$$E_a(k) = C_a \left( \frac{M_g}{M_t} \right) \varepsilon^{1/3} k^{-7/3} \left[ \frac{kL}{[(kL)^2 + C_{aL}]^{1/2}} \right]^{19/3}$$

## SUMMARY

- The next step is to determine the procedure to follow to extract a closure model from

$$\Pi_{ij} = 2\lambda_{lm} (M_{imlj} + M_{jmli})$$

where

$$\lambda_{lm} = S_{lm} + W_{lm}$$

$$M_{ijlm} = \bar{\rho} \left( \frac{M_g}{M_t} \right) \int_0^\infty dk \int_0^\pi \sin \theta d\theta \int_0^{2\pi} d\phi R(k, \theta, \phi; \tau_I) k_l k_m E_{ij}(k, \theta, \phi)$$