

ONERA Large Reference Model in the transonic wind tunnel S1MA - Modane, France

# Wind Tunnel Division



# A wind tunnel for every type of test

Facility	Type Power	Test section L x H	Velocity/ Mach number	Reynolds number based on 0.1VA	Main applications  Optical measurement capabilities: Particle Image Velocimetry Pressure Sensitive Paints Model Deformation Measurement
F1 Fauga-Mauzac	Continuous Pressurized 9.5 MW	Cart 1 4.5 x 3.5 m Cart 2/4 4.5 x 3.5 m Cart 3 4.5 x 3.5 m	M ≤ 0.36 ≤ 123 m/s M ≤ 0.36 123 m/s M ≤ 0.36 ≤ 123 m/s	8 10 <sup>6</sup> 8 10 <sup>6</sup> 8 10 <sup>6</sup>	Model mounts: sting, 3 masts, single mast ½ models with engine simulators Ground effects Air inlets, with or without ground effect PIV, PSP, MDM
F2 Fauga-Mauzac	Continuous Atmospheric 0.75 MW	1.4 x 1.8 m	≤ 100 m/sec	1.1 10 <sup>6</sup>	Research programs Blade profile optimization for wind mills 3-D laser velocimetry with long travel (0.5 x 0.6 x 1 m)  PSP
CEPRA19 Saclay (DGA EP)	Aeroacoustic Continuous Atmospheric 7 MW	Open jet Ø 2 m Open jet Ø 3 m	120 m/s 60 m/s	1.3 10 <sup>6</sup>	Aeroacoustic testing with or without jet simulation  PIV, PSP, MDM
S1MA Modane-Avrieux	Continuous Atmospheric 88 MW	Transonic 3 carts Ø 8 m	M < 1	7 10 <sup>6</sup>	1/2 models, with engine simulators Model mounts: sting or mast Laminar flow tests Air inlets Captive trajectory system Rotating wings, rotors Aeroacoustic measurement capabilities Sting interferences determined with CFD  PSP, MDM
S2MA Modane-Avrieux	Continuous pressurized 55 MW	Transonic 1.75 x 1.77 m Supersonic 1.75 x 1.93 m	M ≤ 1.5 1.5 ≤ M ≤ 3.0	5.4 10 <sup>6</sup> 4.0 10 <sup>6</sup>	Model mounts: sting or wall ½ models with engine simulators Captive trajectories/stores release Air inlets Simulation of hot or cold jets Dynamic stability  PIV, PSP, MDM

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S3MA Modane-Avrieux	Blow down	0.56 x 0.78 m 0.76 x 0.80 m	0.1 ≤ M ≤ 1.3 1.65 ≤ M ≤ 3.8 + fixed nozzles M 2/3.4/4.5/5.5	3.5 10 <sup>6</sup> 3.2 10 <sup>6</sup>	Launchers and missiles  2-D profiles  Mounted on sting or on wall  Precipitation erosion  Air inlets  Jet simulation  PSP
S4MA Modane-Avrieux	Blow down	Ø 0.68 m Ø 1 m Ø 1 m	Hypersonic	0.35-1.7 10 <sup>6</sup>	Load, pressure and thermal flux measurement  Coating thermal and mechanical resistance tests  Aerothermal erosion  Air inlets and ramjets
S4B Modane-Avrieux	Model engine calibration rig	4 " bench 9 " bench	Simulated Mach ≤ 0.9		Thrust/flowrate calibration of flow- through or motorized nacelles  Direct or reverse jet configuration  Laser velocimetry and IR thermographic measurements
BD2 Modane-Avrieux	Dynalpy test bench for nozzle				Flow features :  Primary flow: 16kg/s, T ≤ 1150 K Secondary flow: 14 kg/s, T ≤ 450 K Tertiary flow: 0.7 kg/s  Thrust (13,500 N), side loads, temperatures, pressure and flowrate measurements
R4-1 Modane-Avrieux	Blow down	Ø 600 mm	M < 1		Air intake testing research programs

#### **Quality management, Confidentiality**

Quality Assurance system is certified to be ISO 9001 compliant.

## **Productivity / Availability**

Individual preparation room for model assembly and equipment. Dedicated model carts. Productive testing. High stability in Mach number control.

#### Quality in measurement techniques

New capabilities are permanently developed.

## **Carbon footprint**

S1MA is a water driven wind tunnel.

#### Design office & work shop

In-house model design and manufacture capabilities. Internal balances: 80 balances available and new ones manufactured on request.



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