Propeller powered aircraft *Detailed testing capabilities*

Model type	Model rigging	Wind tunnel	Typical test program	Test objective	Mach number range
Isolated propeller testing. Diameter: 0.5 to 3 m.	Specific test rigs for isolated propeller and installed effect measurements development ongoing.	<u>S1MA</u>	 RPM variations. Blade setting variations. α and β adjustable with fixed wedges. 	 Propeller efficiency. Acoustic measurements (for small models). Pressure measurements (steady and unsteady). 	M<1
Full model. Propeller motorized (with air turbine)	Single strut	<u>F1</u>			M<0,36
Air intake test (with propeller effect)		<u>S1MA</u>	 RPM variations. Blade setting variations. α and β adjustable with fixed wedges. 	 Air intake efficiency. Pressure measurements (steady and unsteady). 	M<1
Full model. Wing span: 3 to 4 m. Propeller motorized (with air turbine)	Straight sting Z sting Fin sting Twin sting	<u>S1MA</u>	. sweep polars (range = 45°). . sweep polars (range = 20°).	 Pressure distribution. Wake survey. Acoustic measurements. Aircraft control qualities. Accurate drag measurements. Buffet onset. Structural loading. Sting near field effect. Additional drag induced by- propellers, 	M<1
Half model Half span: 1 to 4.5 m Propeller motorized (with air turbine)	Bottom wall mounted.	<u>S1MA</u>	α sweep polars. polars with all parameters fixed.	Laminar flow control. Additionnal drag induced by propellers,	M<1